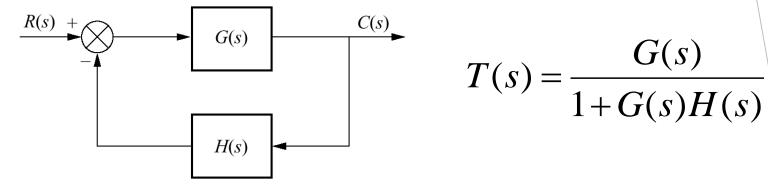
Frequency Response Method

Knowledge Before Studying Nyquist Criterion



unstable if there is any pole on RHP (right half plane)

$$G(s) = \frac{N_G(s)}{D_G(s)} \qquad \qquad H(s) = \frac{N_H(s)}{D_H(s)}$$

Open-loop system:

$$G(s)H(s) = \frac{N_G(s) N_H(s)}{D_G(s) D_H(s)}$$

Characteristic equation:

$$1 + G(s)H(s) = 1 + \frac{N_G N_H}{D_G D_H} = \frac{D_G D_H + N_G N_H}{D_G D_H}$$

poles of G(s)H(s) and 1+G(s)H(s) are the same

Closed-loop system:

$$T(s) = \frac{G(s)}{1 + G(s)H(s)} = \frac{N_G(s)D_H(s)}{D_G(s)D_H(s) + N_G(s)N_H(s)}$$

zero of 1+G(s)H(s) is pole of T(s)

$$G(s)H(s) = \frac{(s-1)(s-2)(s-3)(s-4)}{(s-5)(s-6)(s-7)(s-8)}$$

$$G(s)H(s) = \frac{1+G(s)H(s)}{1+G(s)H(s)} = \frac{G(s)}{1+G(s)H(s)}$$
Zero - 1,2,3,4
$$Zero - a,b,c,d = Zero - ?,?,?,?$$
Poles - 5,6,7,8
$$Poles - 5,6,7,8 = Poles - a,b,c,d$$

To know stability, we have to know a,b,c,d

Stability from Nyquist plot

From a Nyquist plot, we can tell a number of closed-loop poles on the right half plane.

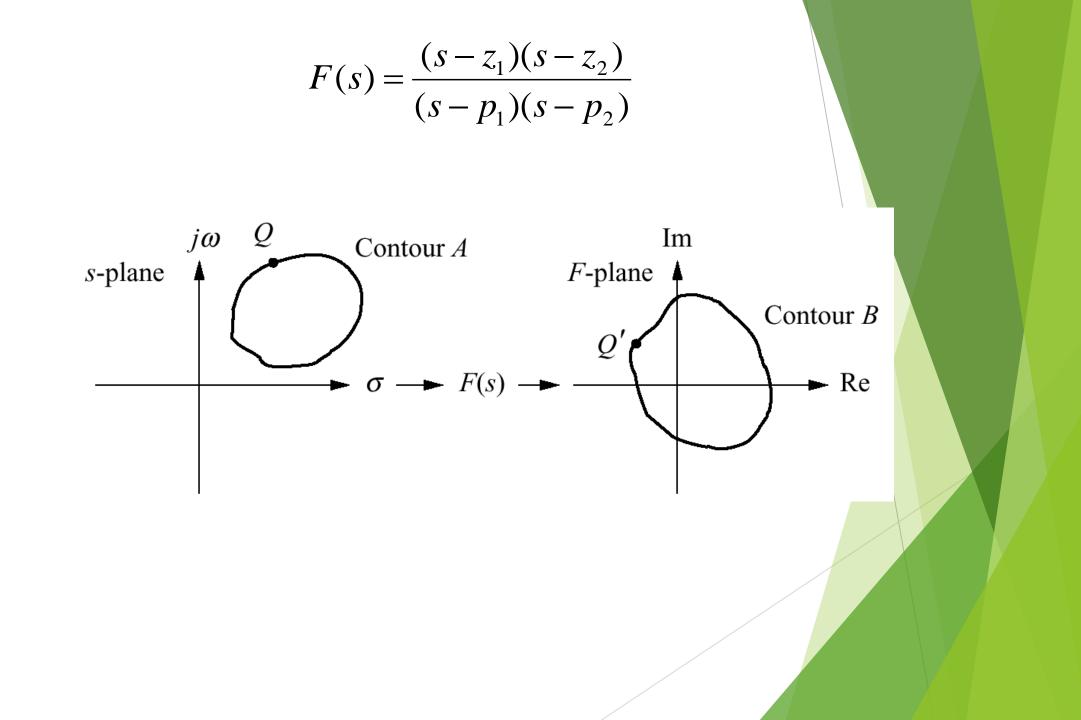
- If there is any closed-loop pole on the right half plane, the system goes unstable.
- If there is no closed-loop pole on the right half plane, the system is stable.

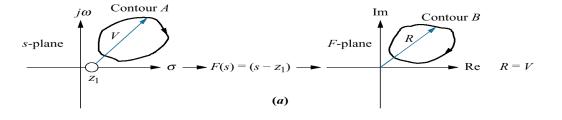
Nyquist Criterion

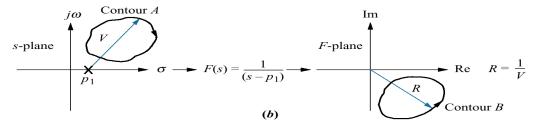
Nyquist plot is a plot used to verify stability of the system.

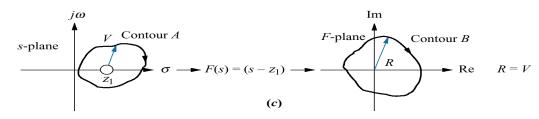
mapping contour function $F(s) = \frac{(s-z_1)(s-z_2)}{(s-p_1)(s-p_2)}$

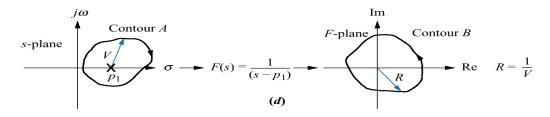
mapping all points (contour) from one plane to another by function F(s).

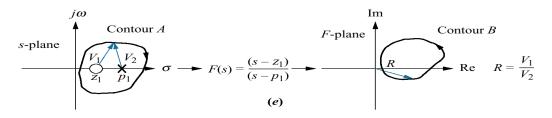








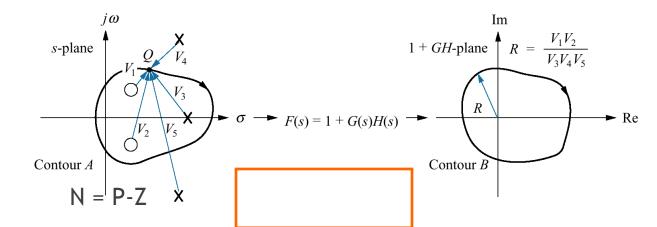




- Pole/zero inside the contour has 360 deg. angular change.
- Pole/zero outside contour has 0 deg. angular change.
- Move clockwise around contour, zero inside yields rotation in clockwise, pole inside yields rotation in counterclockwise

Characteristic equation





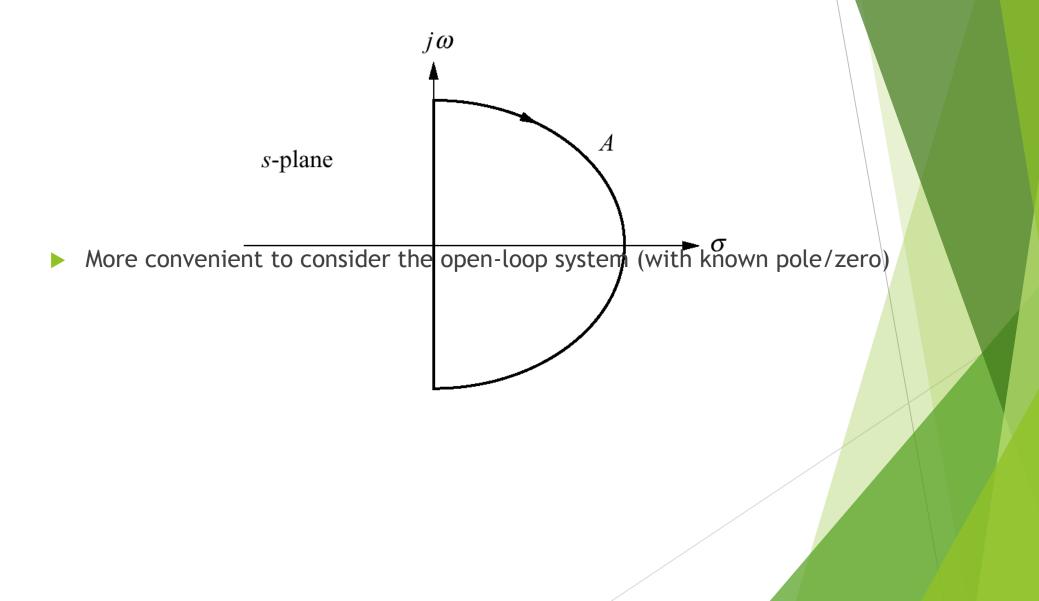
N = # of counterclockwise direction about the origin

- P = # of poles of characteristic equation inside contour
 - = # of poles of open-loop system
- z = # of zeros of characteristic equation inside contour
 - = # of poles of closed-loop system



Characteristic equation

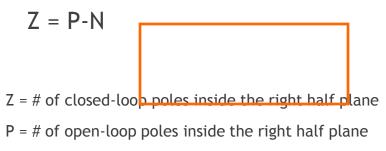
Increase size of the contour to cover the right half plane



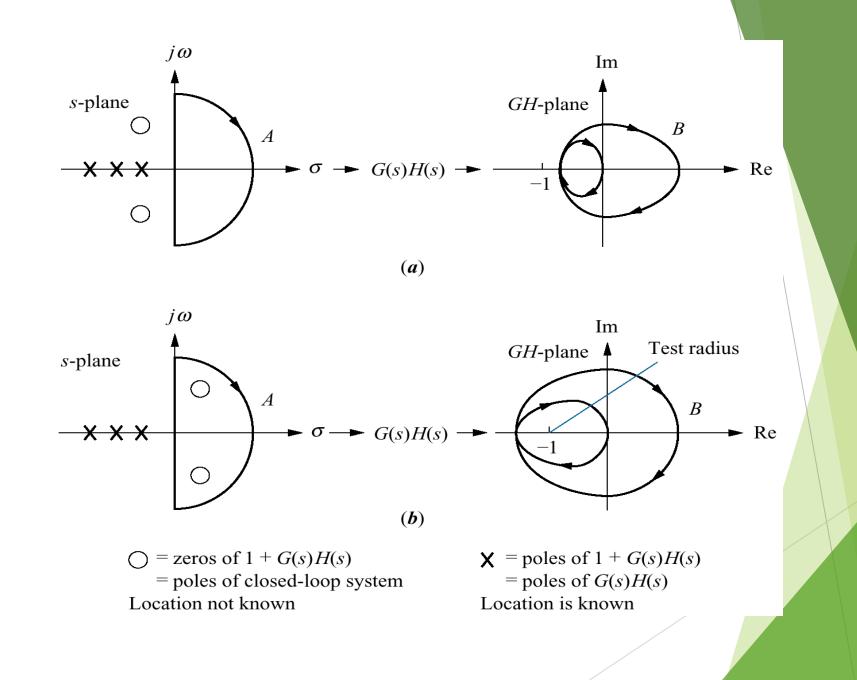
Nyquist diagram of G(s)H(s)

'Open-loop system'

Mapping from characteristic equ. to open-loop system by shifting to the left one step

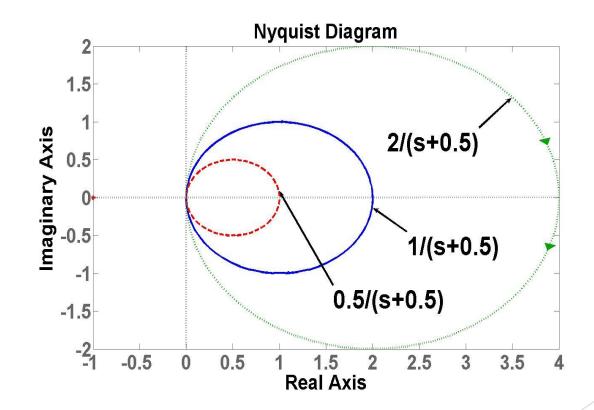


N = # of counterclockwise revolutions around -1

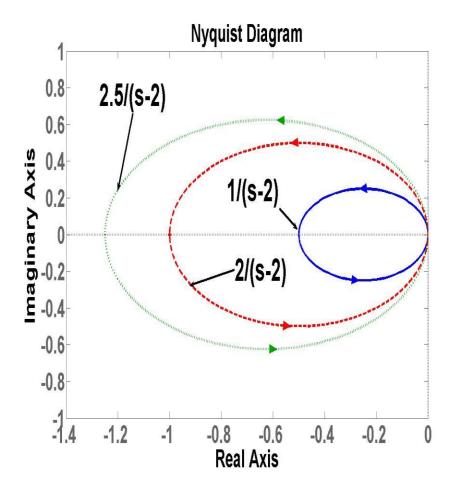


Properties of Nyquist plot

If there is a gain, K, in front of open-loop transfer function, the Nyquist plot will expand by a factor of K.



Nyquist plot example



Open loop system has pole at 2

$$G(s) = \frac{1}{s-2}$$

- Closed-loop system has pole at 1
- $\frac{G(s)}{1+G(S)} = \frac{1}{(s-1)}$ If we multiply the open-loop with a gain, K, then we can move the closed-loop pole's position to the left-half plane

Nyquist plot example (cont.)

New look of open-loop system:

$$G(s) = \frac{K}{s-2}$$
• Corresponding closed-loop system:

• Evaluate value of K for stability
$$I + G(s) = \frac{K}{s + (K-2)}$$

$$K \ge 2$$

Adjusting an open-loop gain to guarantee stability

$$\begin{array}{c|c}
R(s) + & E(s) \\
\hline
K(s+3)(s+5) \\
(s-2)(s-4) \\
\hline
(a)
\end{array}$$

Step I: sketch a Nyquist Diagram Step II: find a range of K that makes the system stable!

How to make a Nyquist plot?

Easy way by Matlab

- Nyquist: 'nyquist'
- Bode: 'bode'

Step I: make a Nyquist plot

Starts from an open-loop transfer function (set K=1)

Set and find frequency response

At dc,
S =
$$j\omega$$

Find at which the imaginary part equals zero
 $\omega = 0 \rightarrow s = 0$
 ω

$$G(s)H(s) = \frac{(s+3)(s+5)}{(s-2)(s-4)} = \frac{s^2 + 8s + 15}{s^2 - 6s + 8}$$

$$G(j\omega)H(j\omega) = \frac{-\omega^2 + 8j\omega + 15}{-\omega^2 - 6j\omega + 8} = \frac{(15 - \omega^2) + 8j\omega}{(8 - \omega^2) - 6j\omega}$$

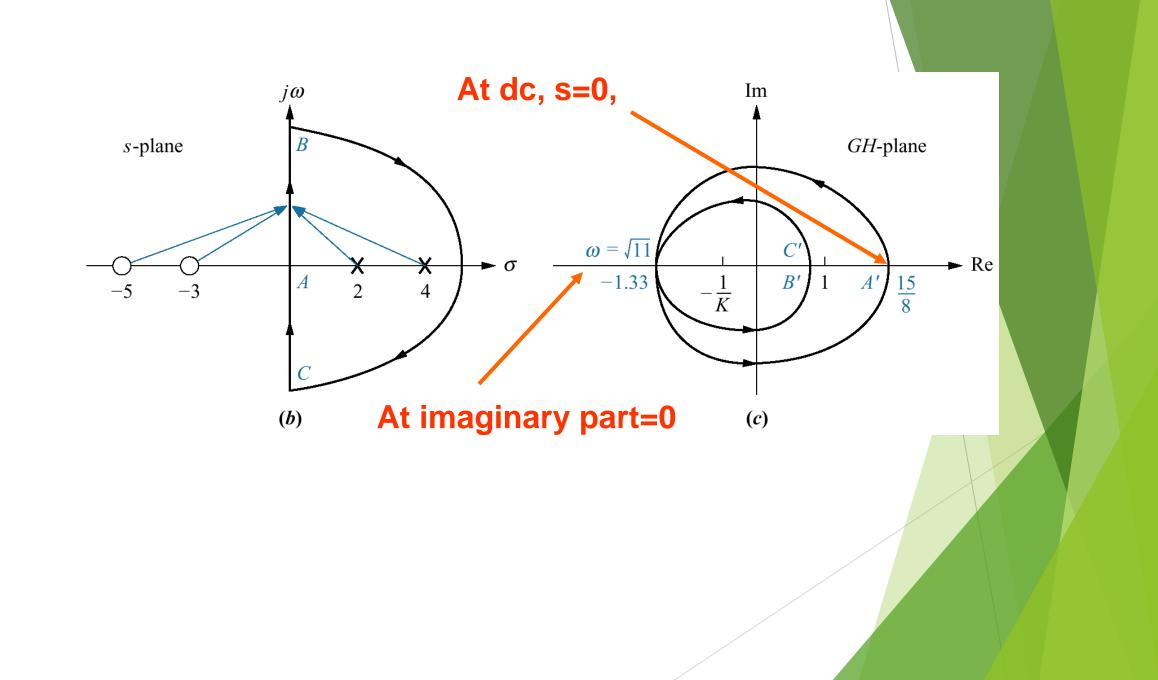
$$= \frac{(15 - \omega^2) + 8j\omega}{(8 - \omega^2) - 6j\omega} \times \frac{(8 - \omega^2) + 6j\omega}{(8 - \omega^2) + 6j\omega}$$

$$= \frac{(15 - \omega^2)(8 - \omega^2) - 48\omega^2 + j(154\omega - 14\omega^3)}{(8 - \omega^2)^2 + 6^2\omega^2}$$

Need the imaginary term = 0,

Substitute $\omega = \sqrt{11}$ back in to the transfer function And get G(s) = -1.33

$$\frac{(15-11)(8-11)-48(11)}{(8-11)^2+6^2(11)} = \frac{-540}{412} = -1.31$$



Step II: satisfying stability condition

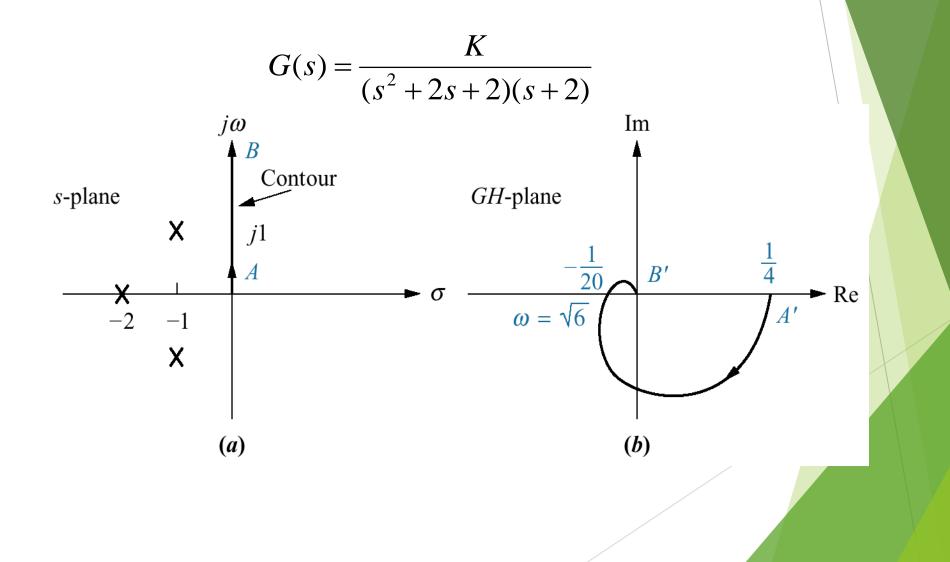
- **P** = 2, N has to be 2 to guarantee stability
- Marginally stable if the plot intersects -1
- **•** For stability, 1.33K has to be greater than 1

K > 1/1.33

or K > 0.75



Evaluate a range of K that makes the system stable



Step I: find frequency at which imaginary part = 0

Set
$$s = j\omega$$

$$G(j\omega) = \frac{K}{((j\omega)^{2} + 2j\omega + 2)(j\omega + 2)}$$
$$= \frac{4(1 - \omega^{2}) - j\omega(6 - \omega^{2})}{16(1 - \omega^{2})^{2} + \omega^{2}(6 - \omega^{2})^{2}}$$

At $\omega = 0, \sqrt{6}$ the imaginary part = 0 Plug $\omega = \sqrt{6}$ back in the transfer function and get G = -0.05

Step II: consider stability condition

- $\blacktriangleright P = 0, N has to be 0 to guarantee stability$
- Marginally stable if the plot intersects -1
- **For stability, 0.05K has to be less than 1**

K < 1/0.05

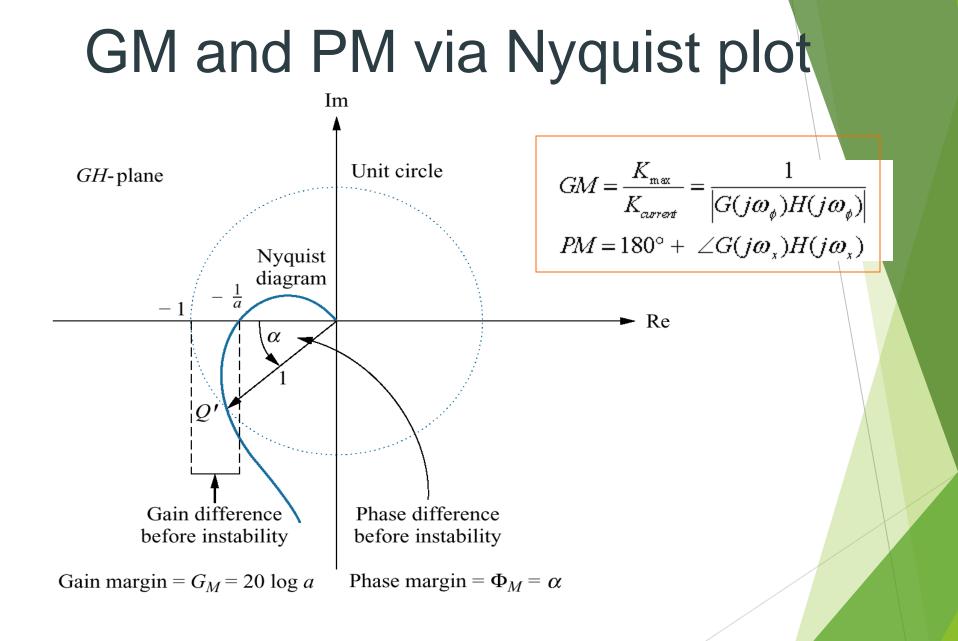


Gain Margin and Phase Margin

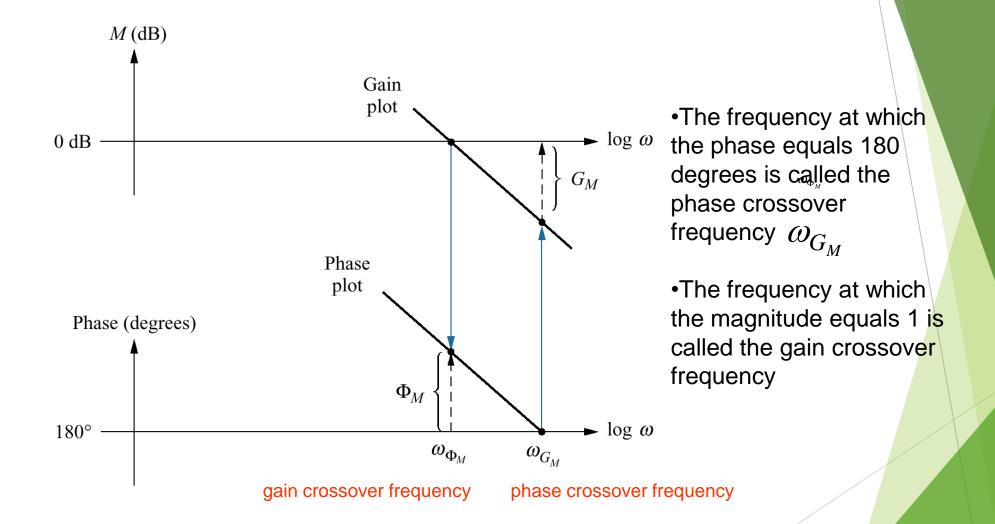
Gain margin is the change in open-loop gain (in dB), required at 180 of phase shift to make the closed-loop system unstable.

Phase margin is the change in open-loop phase shift, required at unity gain to make the closed-loop system unstable.

GM/PM tells how much system can tolerate before going unstable!!!



GM and PM via Bode Plot

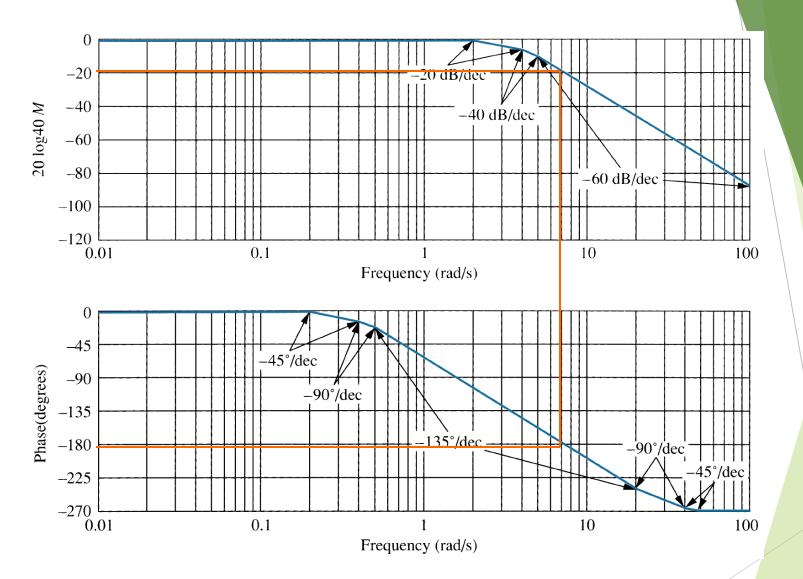


Example

Find Bode Plot and evaluate a value of K that makes the system stable The system has a unity feedback with an open-loop transfer function

$$G(s) = \frac{K}{(s+2)(s+4)(s+5)}$$

First, let's find Bode Plot of G(s) by assuming that K=40 (the value at which magnitude plot starts from 0 dB)



At phase = -180, ω = 7 rad/sec, magnitude = -20 dB

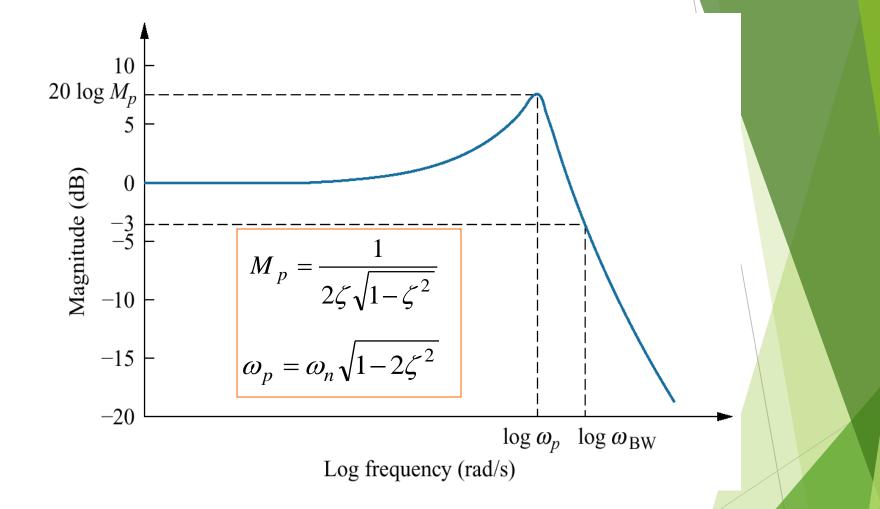
- ► GM>0, system is stable!!!
- Can increase gain up 20 dB without causing instability (20dB = 10)
- **Start from K = 40**
- with K < 400, system is stable</p>

Closed-loop transient and closed-loop frequency responses '2nd system'

$$\frac{R(s) + E(s)}{s(s+2\zeta\omega_n)} \qquad C(s)$$

$$\frac{C(s)}{R(s)} = T(s) = \frac{\omega_n^2}{s^2 + 2\zeta\omega_n s + \omega_n^2}$$

Damping ratio and closed-loop frequency response



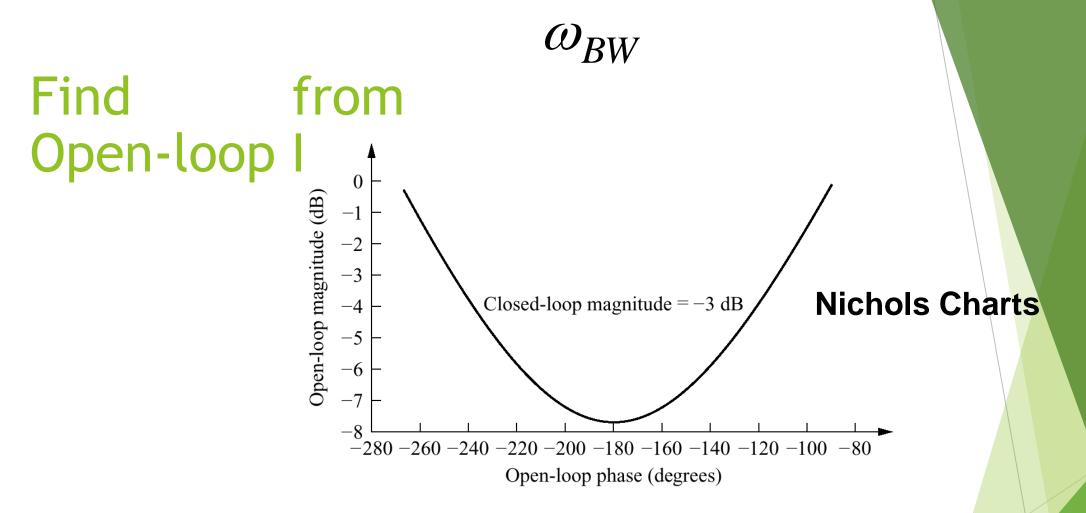
Magnitude Plot of closed-loop system

Response speed and closed-loop frequency response

$$\begin{split} \omega_{BW} &= \omega_n \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}} \\ \omega_{BW} &= \frac{4}{T_s \zeta} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}} \\ \omega_{BW} &= \frac{\pi}{T_p \sqrt{1 - \zeta^2}} \sqrt{(1 - 2\zeta^2) + \sqrt{4\zeta^4 - 4\zeta^2 + 2}} \end{split}$$

 ω_{BW} = frequency at which magnitude is 3dB down

from value at dc (0 rad/sec), or $M = \frac{1}{\sqrt{2}}$.



From open-loop frequency response, we can find ω_{BW} at the open-loop frequency that the magnitude lies between -6dB to -7.5dB (phase between -135 to -225)

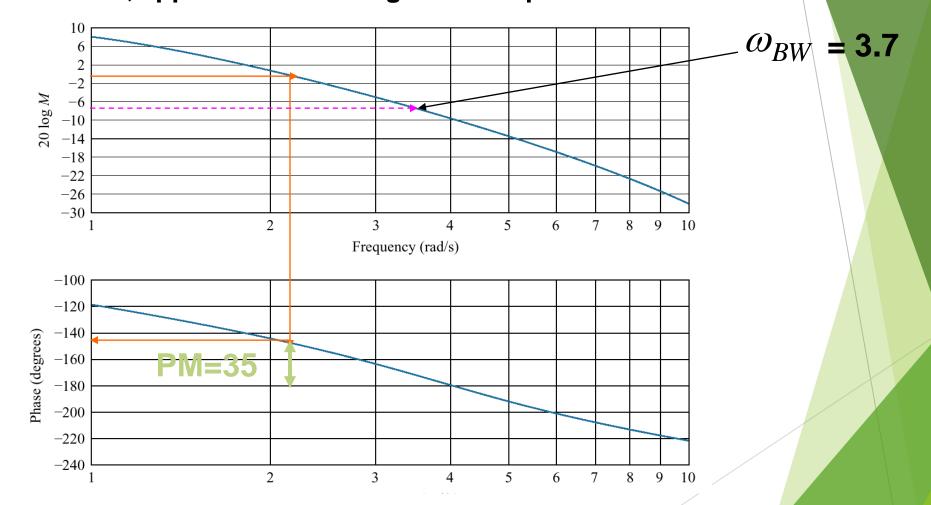
Relationship between damping ratio and phase margin of open-loop frequency response

Phase margin of open-loop frequency response Can be written in terms of damping ratio as following

$$\phi_M = \tan^{-1} \frac{2\zeta}{\sqrt{-2\zeta^2 + \sqrt{1 + 4\zeta^4}}}$$

Example

Open-loop system with a unity feedback has a bode plot below, approximate settling time and peak time



$$\phi_M = \tan^{-1} \frac{2\zeta}{\sqrt{-2\zeta^2 + \sqrt{1 + 4\zeta^4}}}$$

Solve for PM = 35 $\zeta = 0.32$

$$T_{s} = \frac{4}{\omega_{BW}\zeta} \sqrt{(1 - 2\zeta^{2}) + \sqrt{4\zeta^{4} - 4\zeta^{2} + 2}}$$

= 5.5
$$T_{p} = \frac{\pi}{\omega_{BW}\sqrt{1 - \zeta^{2}}} \sqrt{(1 - 2\zeta^{2}) + \sqrt{4\zeta^{4} - 4\zeta^{2} + 2}}$$

= 1.43